

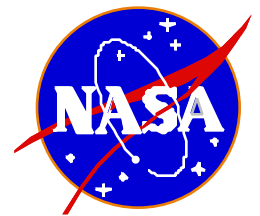
NEXTOR Annual Research Symposium

November 14, 1997

Session III
Issues for the Future of ATM

CTAS Verification
Darren Cofer, Honeywell

Formal Specification and Analysis of the Center-TRACON Automation System



National Center of Excellence for Aviation Operations Research
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Project sponsor: NASA Langley Research Center

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Team

- **Honeywell Technology Center**
Darren Cofer, Rosa Weber, John Maloney
- **University of California at Berkeley**
George Pappas, Shankar Sastry
- **Massachusetts Institute of Technology**
John Lygeros, Nancy Lynch

Unique capabilities and emphasis

- Modeling/analysis of complex systems

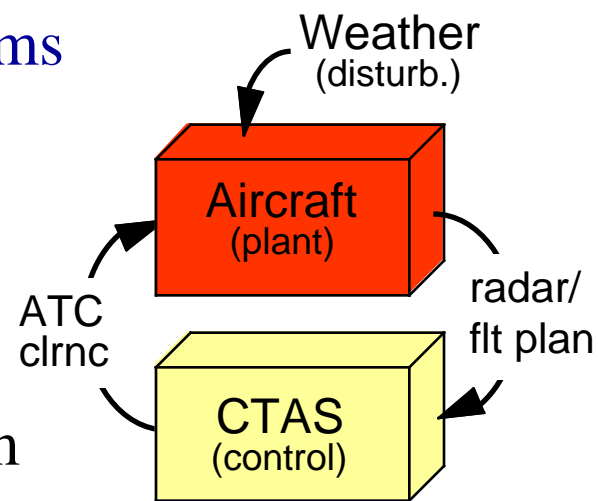
HW + SW architecture issues, tools

- Control-theoretic viewpoint

CTAS as input/output control system

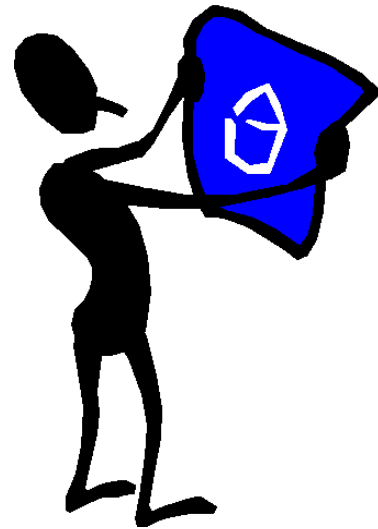
- Hybrid systems

Discrete-event and continuous dynamics

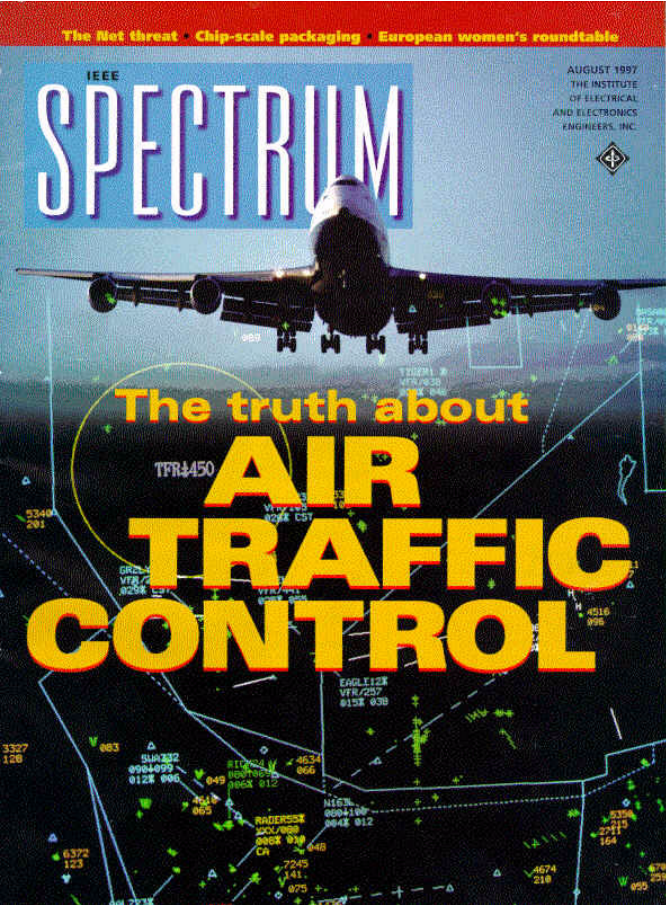


Roadmap for presentation

- What CTAS means to me...
- Changes in ATM/NAS
- Safety Issues and Technologies
- Requirements Specification: *HOPTs*
- System Architecture: *MetaH*
- Formal Systems Analysis: *Hybrid Systems*



Assessing changes in NAS



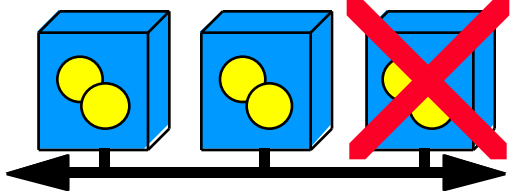
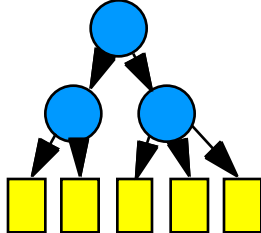
- Increasing demands on system
- New technologies
- New procedures



Impact on system?
Affected components?
Safety?

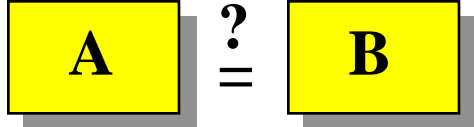
Safety issues

Logical correctness of requirements and implementation

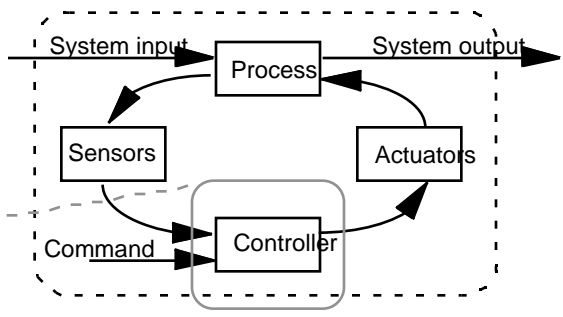


System architecture and timing, degraded modes of operation

Formal equivalence of systems, preservation of properties



Operationally embedded reactive systems



- Modes / Procedures
- Ascent
 - Descent
 - Level Cruise
 - Emerg. procedures
 - Conflict res. mode

- Altitude target
- Speed target
- Vertical speed target
- Pitch/thrust control mode

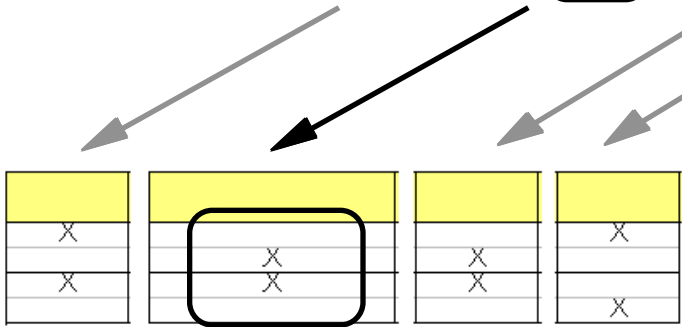
Semantics

1. Get input state

Inputs	States
Prev Op Proc	climb
	altitude capture
	altitude hold
	descend
new_ref_altitude	yes
	no
altitude	greater than ref
	within ref
	less than ref

climb	altitude capture	altitude hold	descend
	lost ref altitude	new altitude	
		X	
			X
X	X		X
		X	
	X	X	
	X	X	X
X	X	X	

2. Compute scenarios and select operation procedure



3. Collect associated behavior output functions

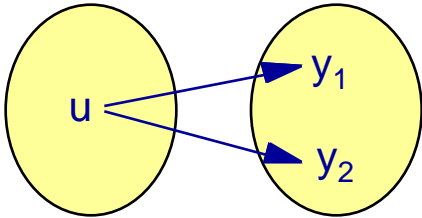
Outputs	Functions
thrust	altitude control
	speed control
elevator	altitude control
	AOA control

4. Execute

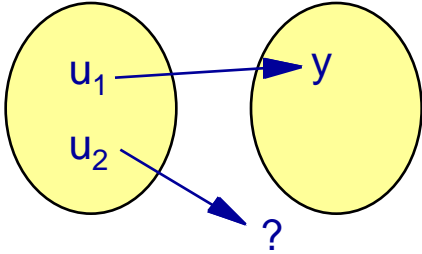
Completeness and Consistency

$$f : U \rightarrow Y$$

- **Consistent:** $f(u)$ is unique.
(a *function* vs. a relation)

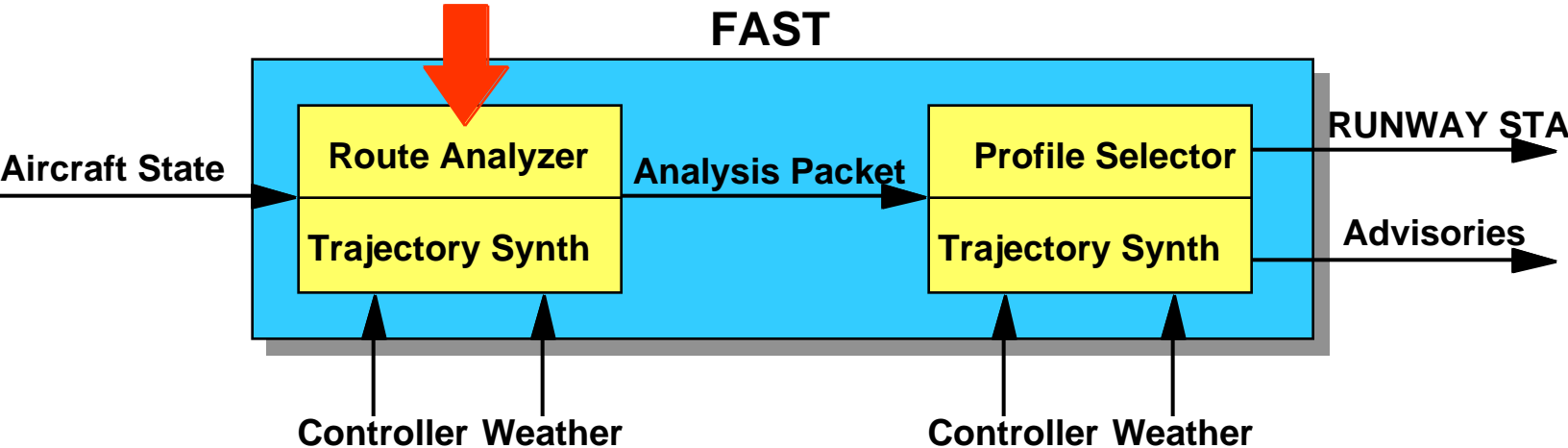


- **Complete:** $f(u)$ defined $\forall u \in U$.
(a *total* function)



CTAS decision logic

Focus on algorithms
in Route Analyzer



Route Analyzer

Contains many decision logic elements

Title:

Creator: DoME by Honeywell Technology Center, Honeywell Inc.

CreationDate:

Update heading (OPT)

Title:

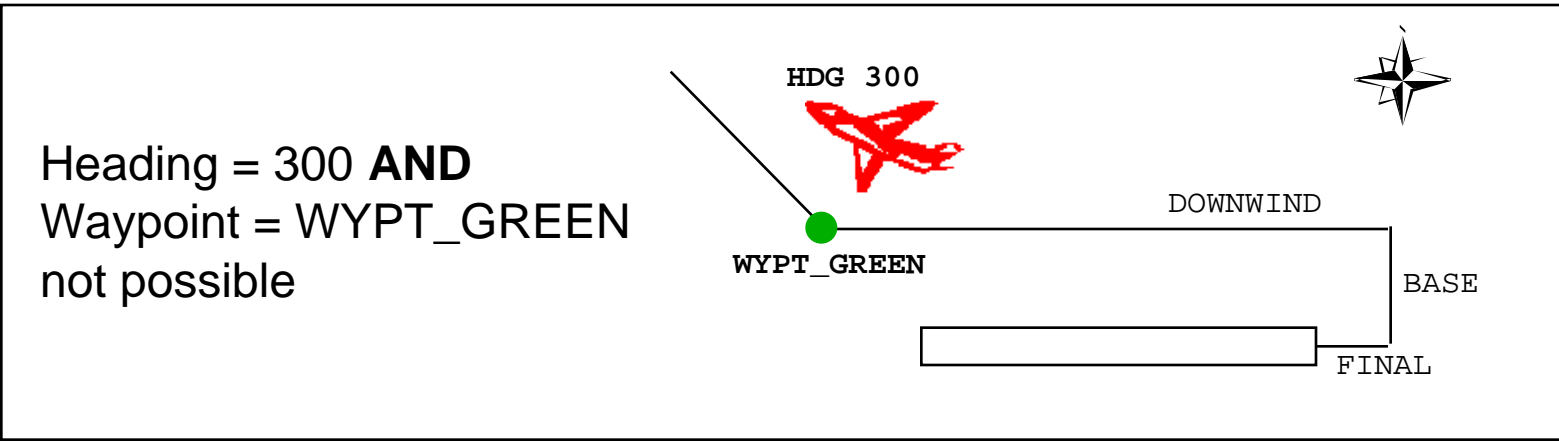
Creator: DoME by Honeywell Technology Center, Honeywell Inc.

CreationDate:

Findings

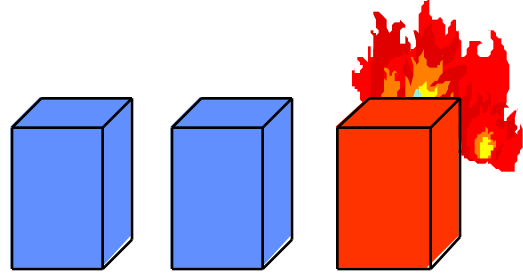
- Consistent
- Some incomplete

May rely on context:

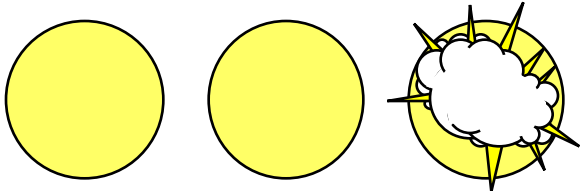


Assess performance in degraded operational modes

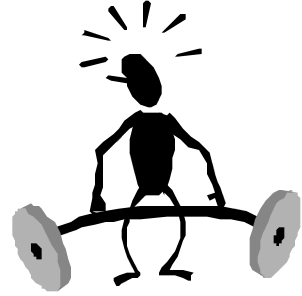
- Hardware node failure



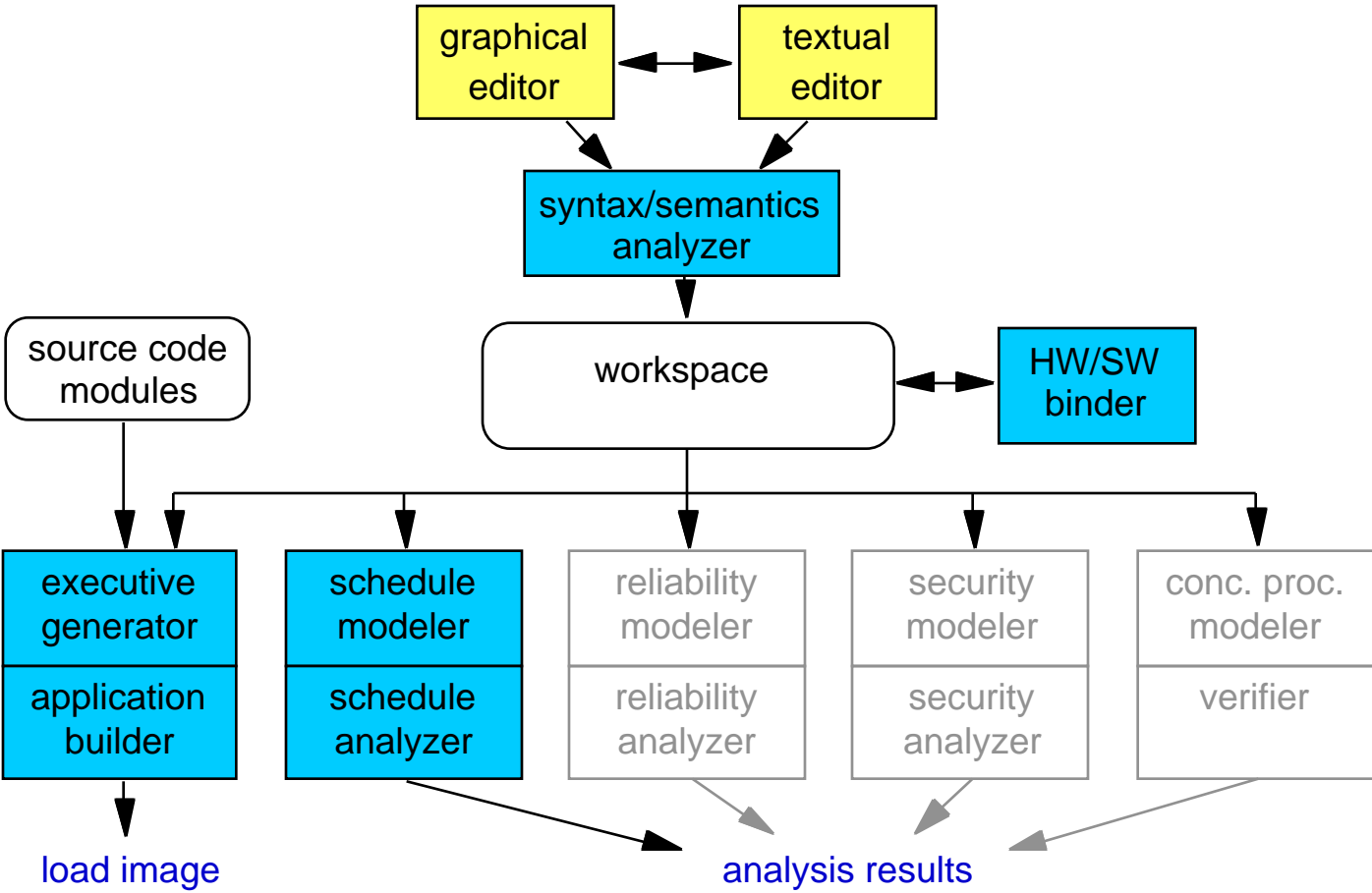
- Software failure



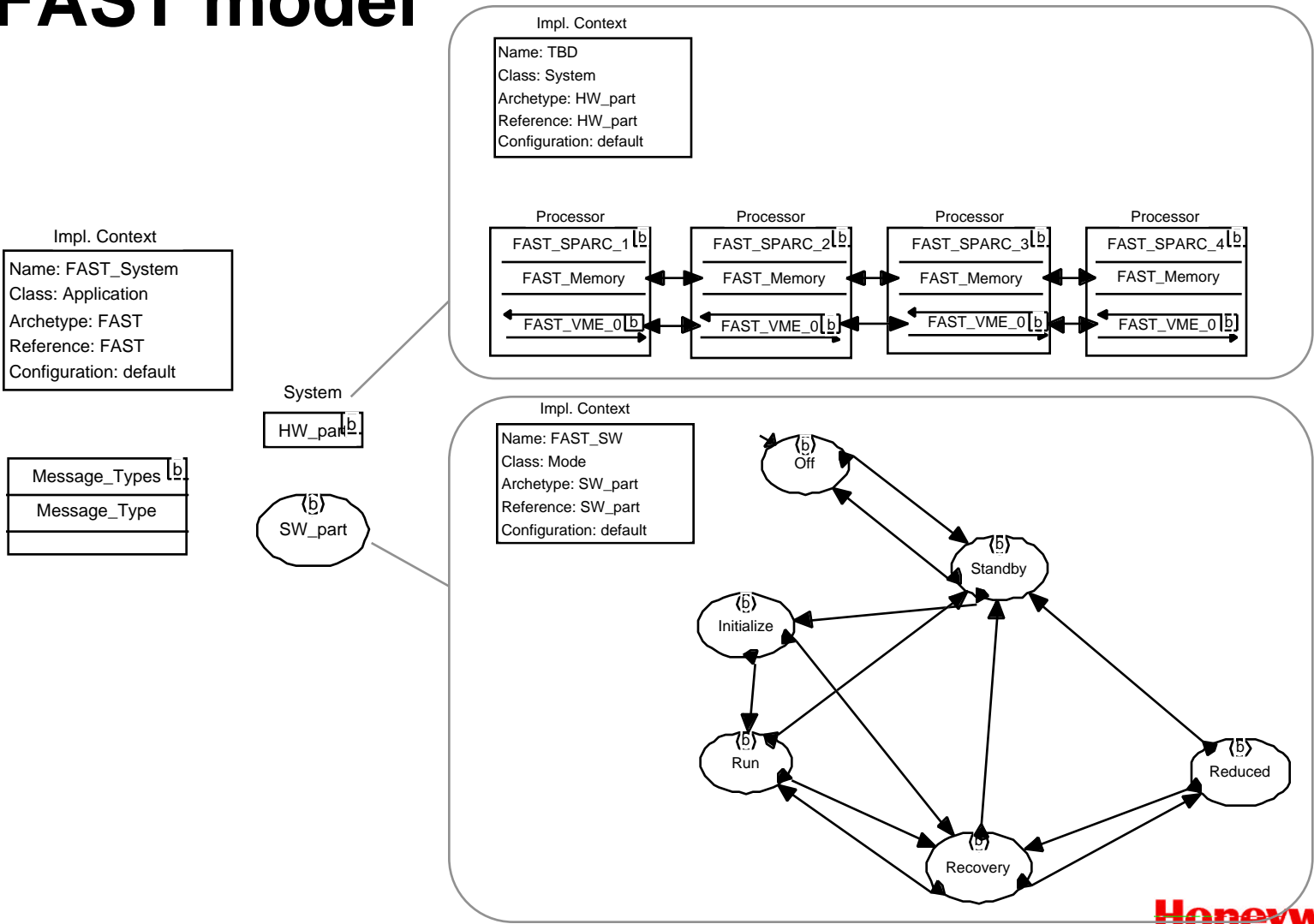
- Excess computational load



What is MetaH?



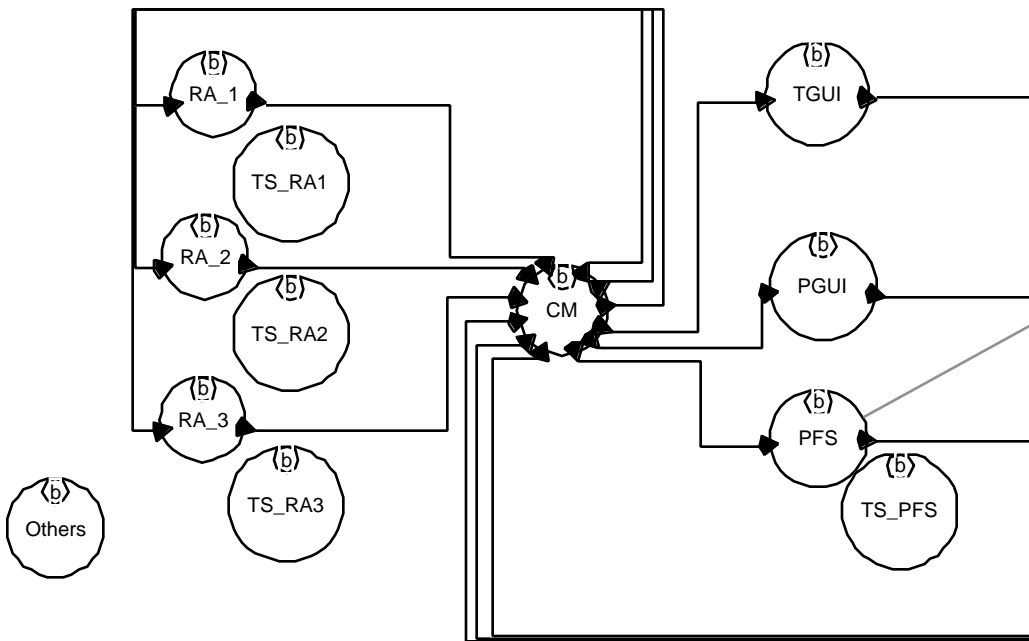
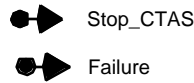
FAST model



FAST model: Run mode

Impl. Context

Name: Run_imp
 Class: Mode
 Archetype: Run
 Reference: Run
 Configuration: default



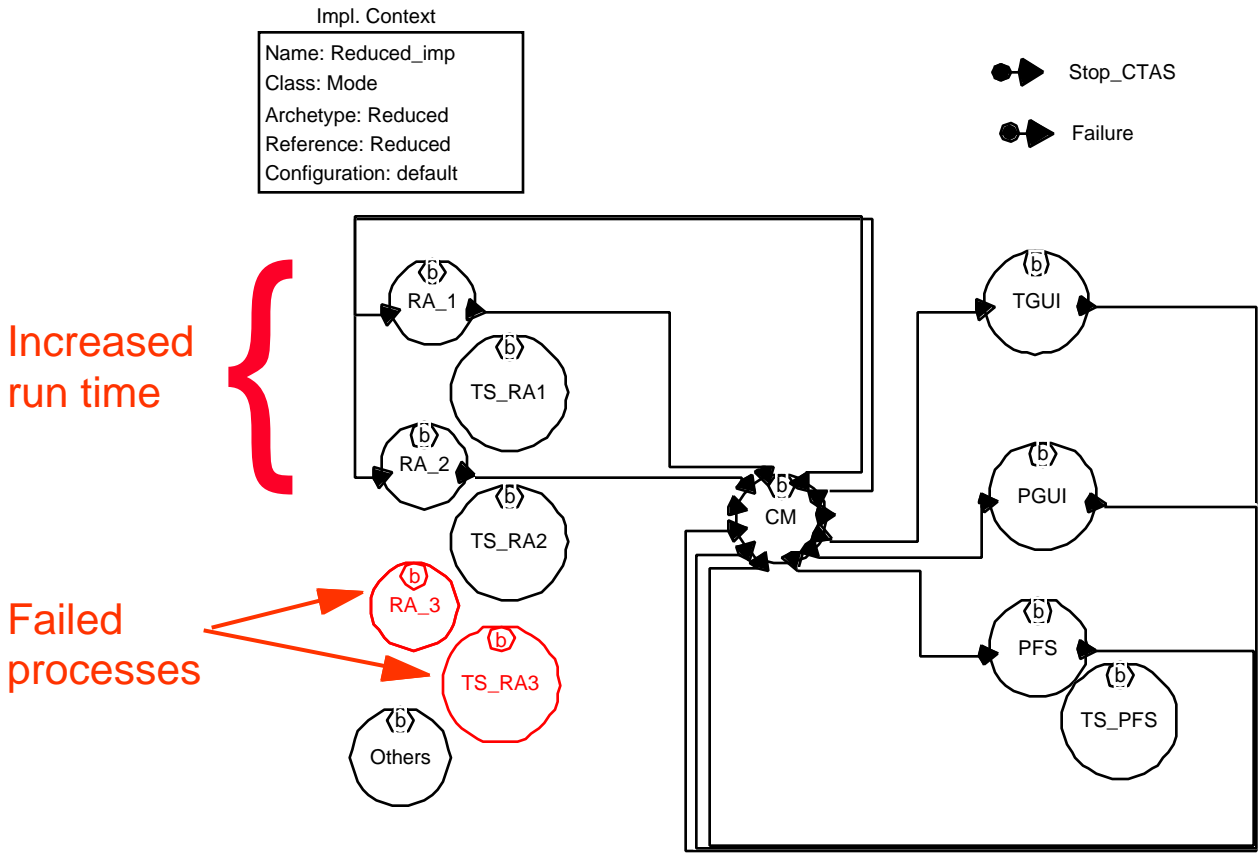
Inspector Edit Category

Name: PFS

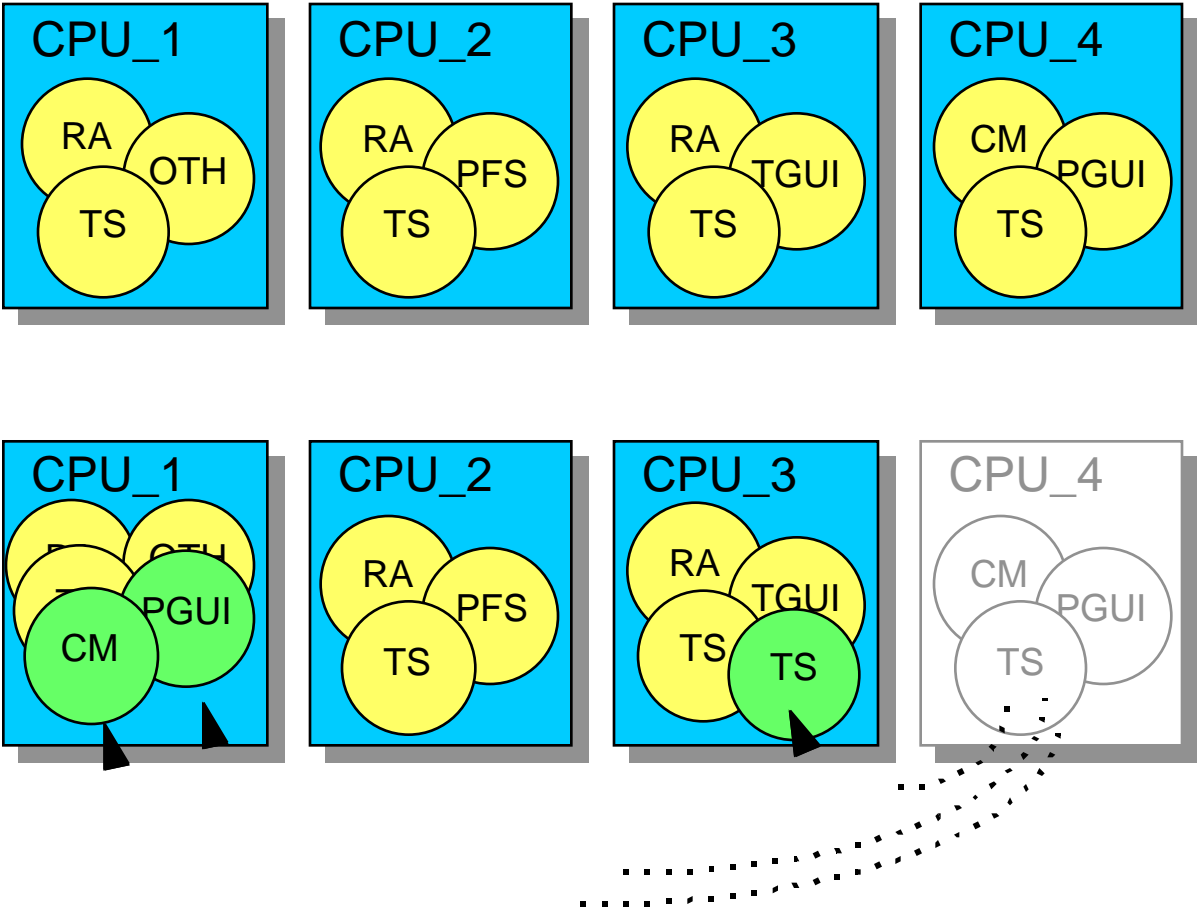
Name	Description	Rationale	Traceability	Color
X-Refs	Overlays	Attributes	Declaration	Properties
Periodicity: <input checked="" type="checkbox"/> periodic				

Apply Close

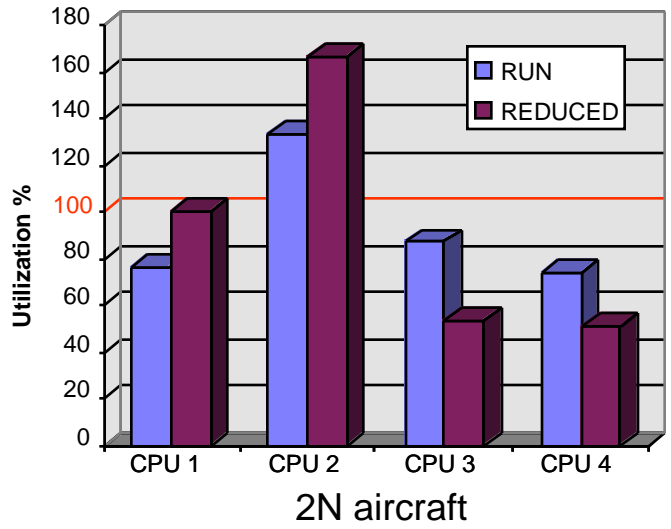
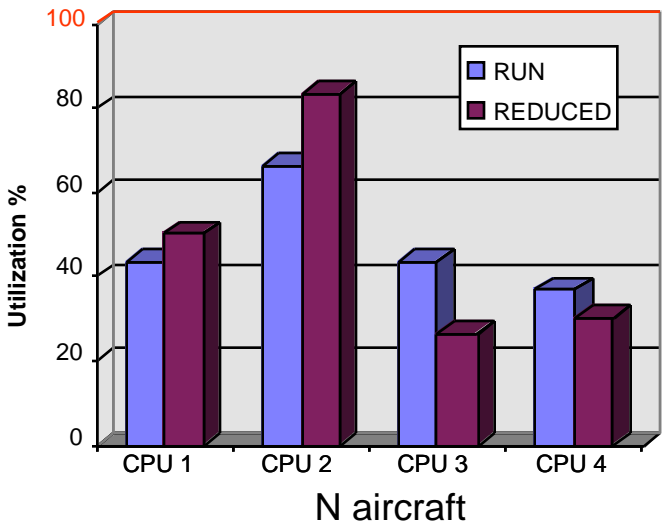
FAST Model: Reduced mode



FAST model: HW-SW binding

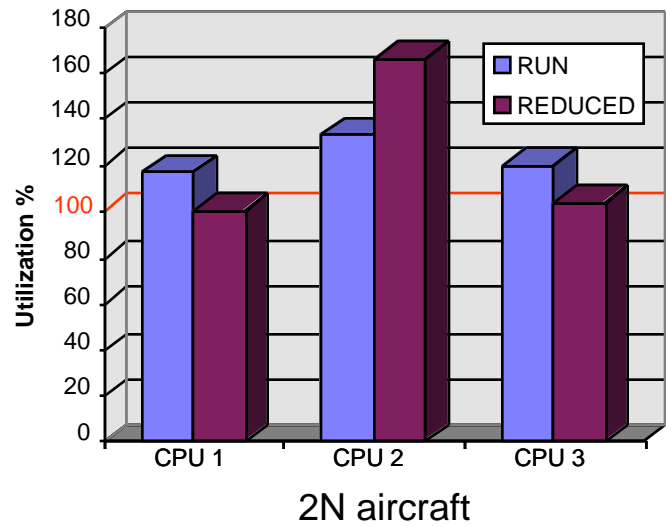
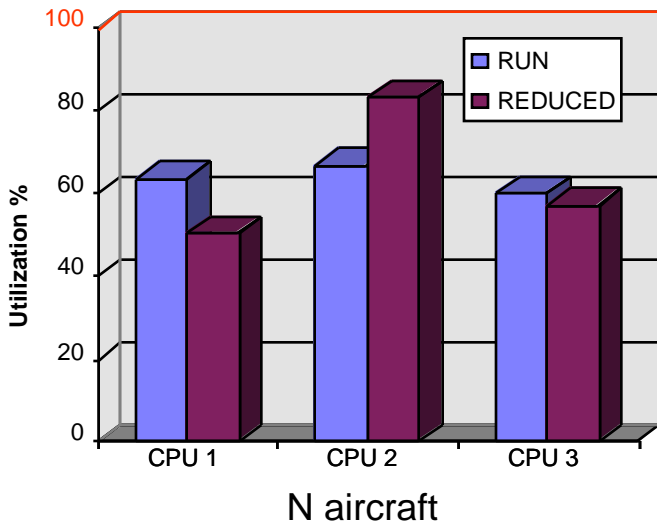


Performance: Nominal Hardware



- In Reduced mode, processor load becomes more unbalanced. Reduces margin to schedulability
- Doubling number of aircraft results causes scheduling failure. Unable to meet deadline for updates.

Performance: One node failure

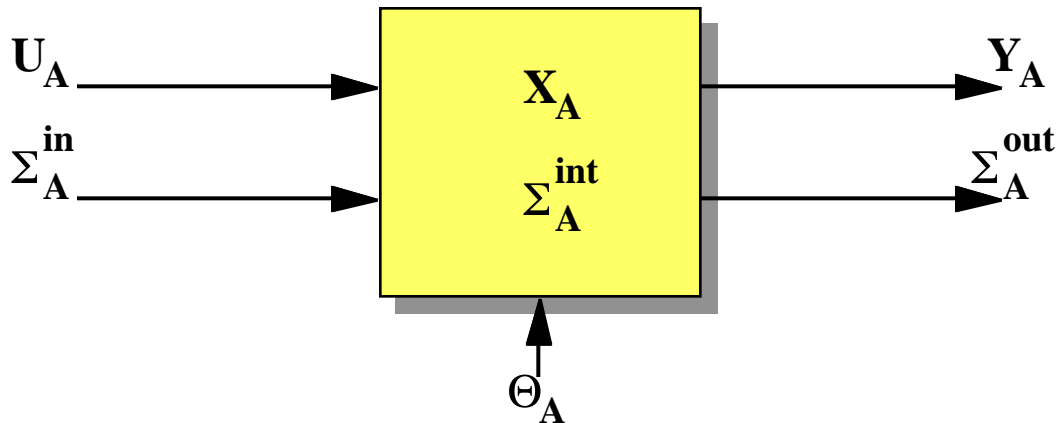


- Load from failed CPU transferred to less busy nodes.
Reduces margin to schedulability on those nodes.
- Doubling number of aircraft results in scheduling failure.
Unable to meet deadline for updates.

Other analyses...

- New processes added to system
 - Departure automation
- New capabilities added to existing processes
 - Weather data in route analysis
- Faster cycle times required
 - Fast radar updates or GPS data

Hybrid Input/Output Automata



A hybrid input/output automaton A is defined by

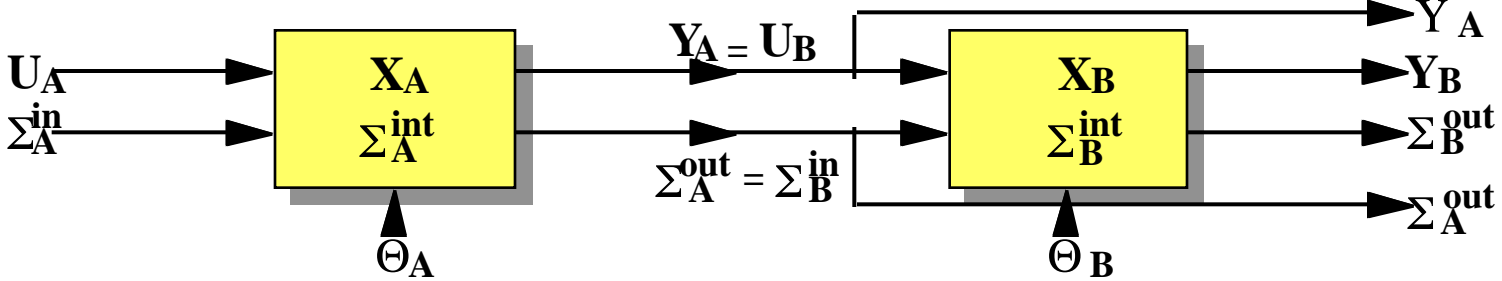
- Input, output and internal typed variables
- Input, output and internal actions
- State space is set of all possible variable values
- Initial conditions
- A set W of trajectories of variables and D of discrete transitions

Each action has an associated precondition and effect

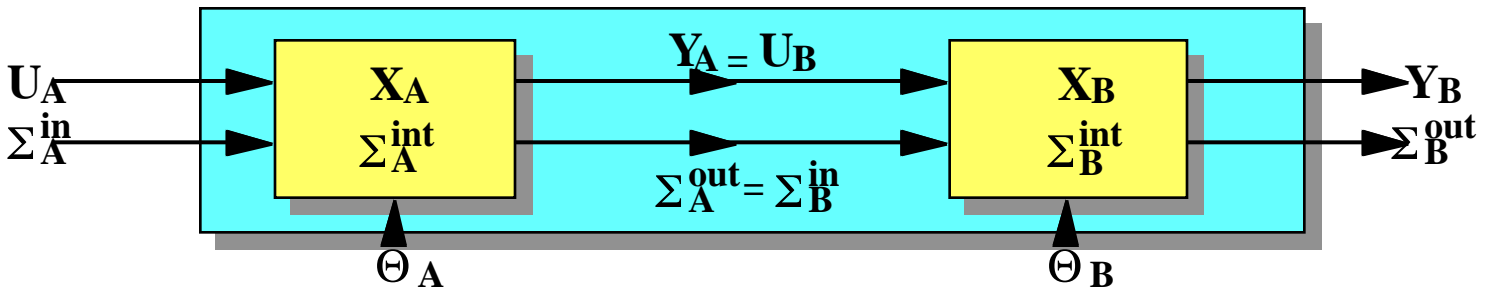
An execution of the automaton is = **w1 a1 w2 a2 w3 a3.....**

Hybrid Input/Output Automata

Compositions of compatible hybrid automata are hybrid automata



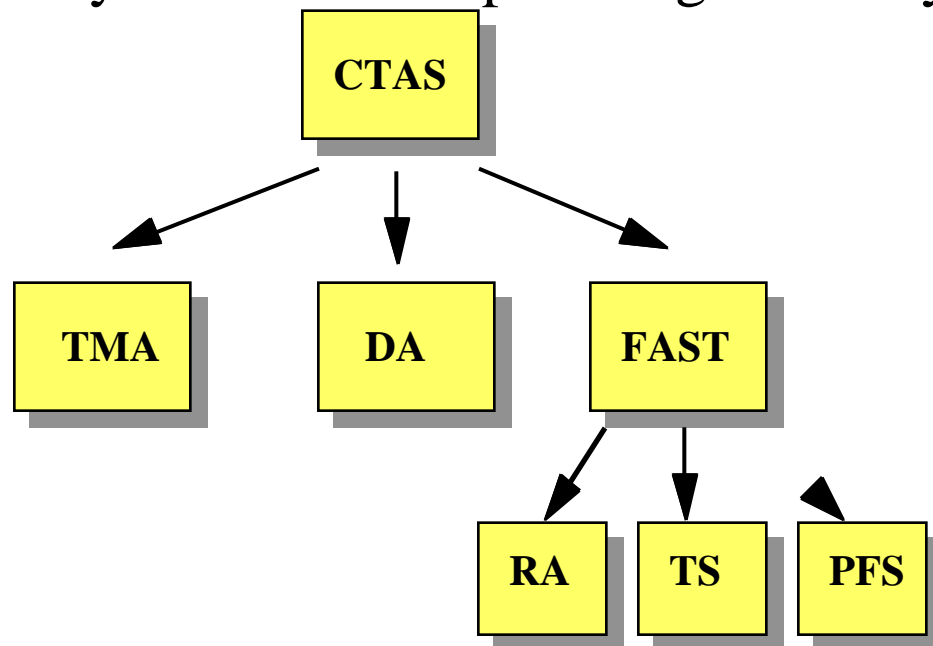
- Variable and action hiding allows building macrocomponents



- Composite system satisfies composite specification

Safety Analysis

How can one analyze such a complex large scale system?



- Step 1 : Top down specification refinement
- Step 2 : Verify that low level systems meet specification
- Step 3 : Abstract behavior of composite system

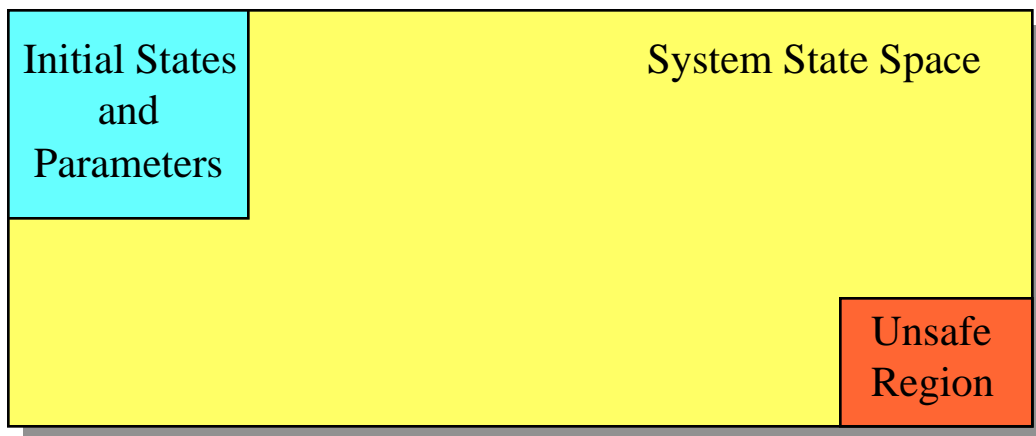
Safety Analysis

Safety specs can be expressed as undesirable state regions

Will aircraft lose separation? Is TRACON capacity exceeded?

Specs can also be formulated using performance monitors

The analysis approach: Forward & Backward Reachability



- **Forward** : *Verify safety given parameters and initial states or generate trajectory leading to unsafe operation*
- **Backward** : *Determine which initial states and parameters are reachable from the unsafe region*

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Safety Tools

Discrete Systems

COSPAN (Correctness of communication protocols)

VIS (Correctness of hardware/software systems)

Timed Systems

KRONOS (real-time properties of communication networks)

Timed COSPAN

Hybrid Systems

HyTech (Rectangular Hybrid Systems)

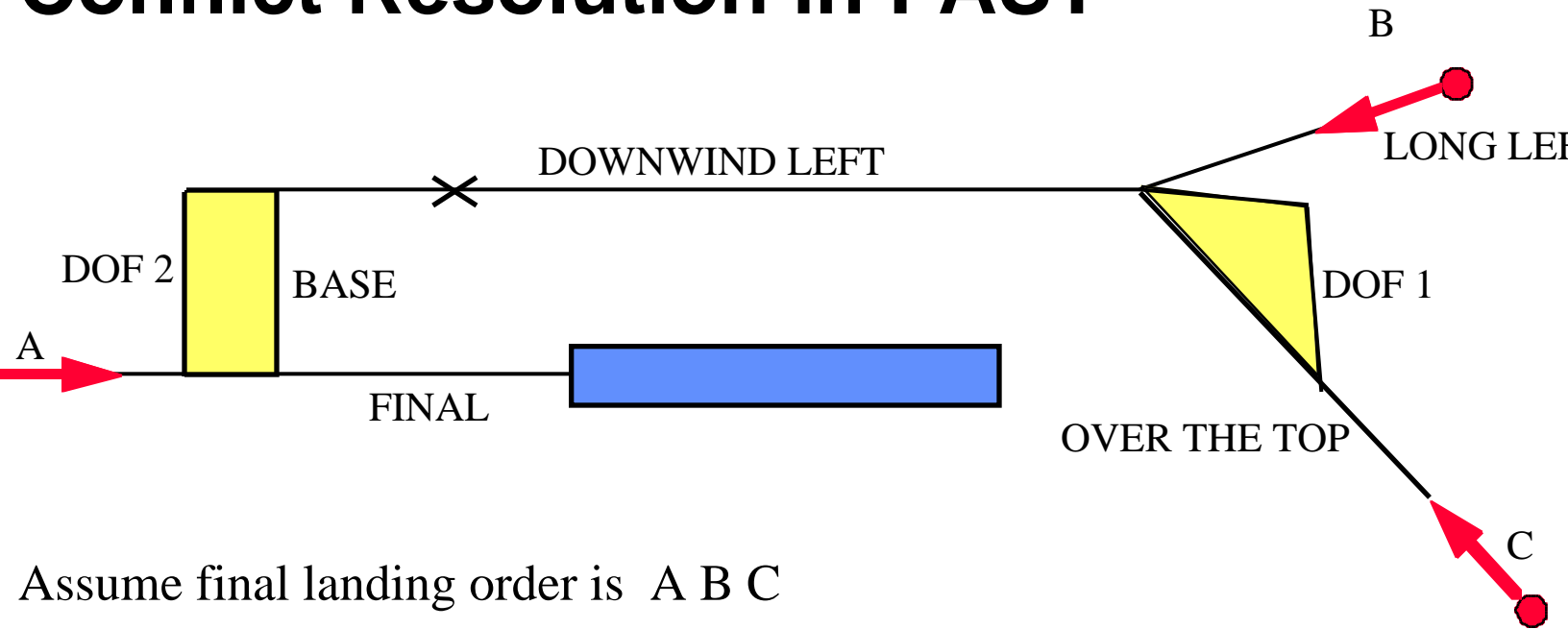
Various Mathematical Tools from

Systems Theory

Probability Theory

Computer Science and Logic

Conflict Resolution in FAST



Assume final landing order is A B C

Potential conflict between B and C on downwind left

Aircraft C must be delayed using 2 degrees of freedom

Speed and altitude profiles dictated by TRACON procedures

Question: For what initial configurations (horizontal and vertical coordinates) of Aircraft C is conflict avoided?



Conclusions

System perspective of safety analysis

Formal Methods Approach

Modeling, specification and analysis

Safety assessment of NAS is similar conceptually

Methodology does not depend on CTAS details

Questions are challenging, but are the right ones!